



National Doctoral Programme in Space Science and Technology : 2024 PhD Days



Gossamer technologies for human and robotic space exploration

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Politecnico di Torino, Torino, Italy

7 June 2024

Gran Sasso Science Institute, L'Aquila, Italy





Agenzia
Spaziale
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**Space It Up un'opportunità
eccezionale per la comunità italiana
'spazio'**

Erasmo Carrera
Politecnico di Torino





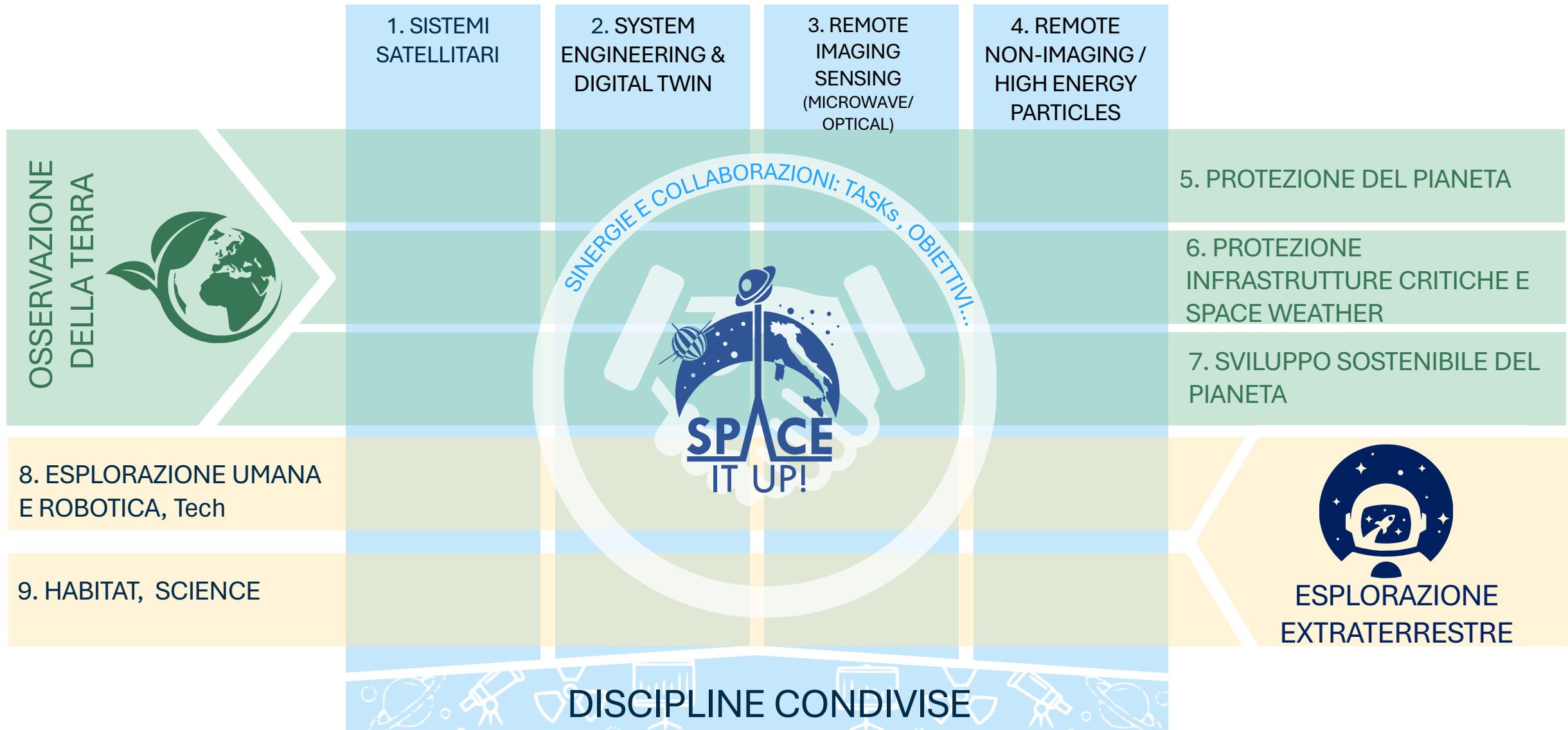
Il progetto Space It Up, finanziato dall'**Agenzia Spaziale Italiana (ASI)** e **dal Ministero dell'Università e della Ricerca (MUR)**, riunisce le competenze italiane nella scienza e nell'ingegneria spaziale, ponendo l'Italia all'avanguardia nella ricerca sull'osservazione e la protezione della Terra, l'esplorazione extraterrestre, i satelliti artificiali e il telerilevamento.



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9 Spokes



33 Partners: 13 Universities, 10 Research Center, 10 Enterprise



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POLITECNICO
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UNIVERSITÀ DEGLI STUDI DI NAPOLI
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INAF
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DI RICERCA METROLOGICA

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ENEA FDK
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FUTURE BUILT
ON KNOWLEDGE

FONDATION
links
PASSION FOR INNOVATION

LEONARDO

ThalesAlenia
Space
a Thales / Leonardo company

TELESPAZIO
a LEONARDO and THALES company

ALTEC

e-geos
AN ASI / TELESPAZIO COMPANY

CIRA
Centro Italiano Ricerche Aerospaziali

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Tyvak International
A Terran Orbital Corporation

MAPSAT
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DI TORINO



Università degli Studi "G. d'Annunzio"
CHIETI - PESCARA



UNIVERSITÀ
DI PARMA

Overview

Extended Partnership: **SPACE IT UP!**
Duration of the program: **30 months**
Cost of the program: **80M€**



PROJECT FIGURES

PARTNERS	33	ENTERPRISES & SME
UNIVERSITIES	13	KEY EXPORTABLE RESULTS
RESEARCH CENTERS	10	SPOKES
NEW RESEARCH FELLOWS	180+	PhD POSITIONS

- 1. ENABLING TECHNOLOGIES FOR NOVEL NEAR-EARTH AND EXPLORATION MISSIONS
- 2. ADVANCED DESIGN AND ANALYSIS OF SPACE MISSIONS AND SYSTEMS AND INNOVATIVE DIGITALIZATION - SYSTEM ENGINEERING AND DIGITAL TWIN
- 3. FUTURE IMAGING SYSTEMS FOR MICROWAVE AND OPTICAL REMOTE SENSING
- 4. REMOTE NON-IMAGING/HIGH ENERGY PARTICLES



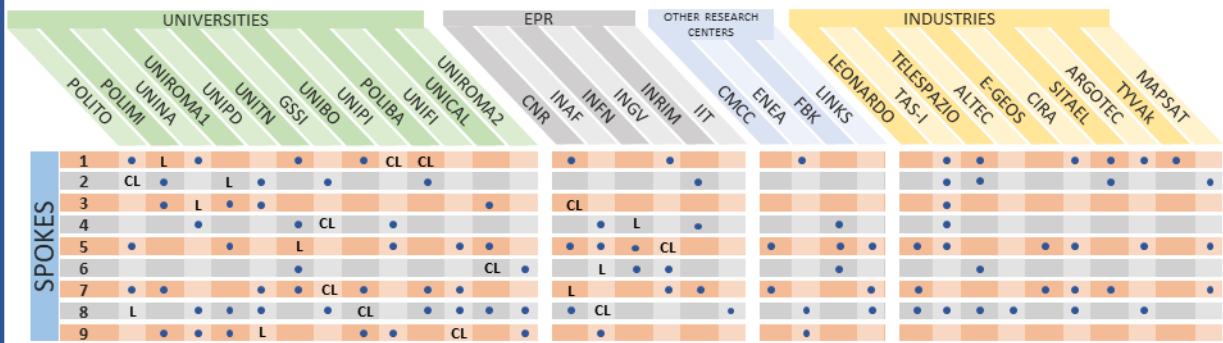
- 5. PLANETARY PROTECTION AND GEOHAZARDS MITIGATION
- 6. PROTECTION OF CRITICAL INFRASTRUCTURES AND SPACE WEATHER
- 7. SPACE FOR THE SUSTAINABLE DEVELOPMENT OF THE PLANET



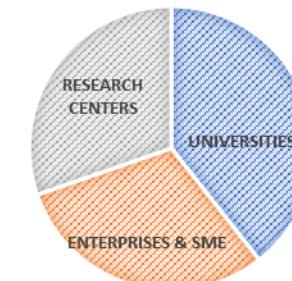
8. ROBOTIC AND HUMAN EXPLORATION OF EXTRATERRESTRIAL HABITATS, ARCHITECTURES AND INFRASTRUCTURES

9. HABITAT SPACE AND SCIENCE

FUNDAMENTAL KNOWLEDGE



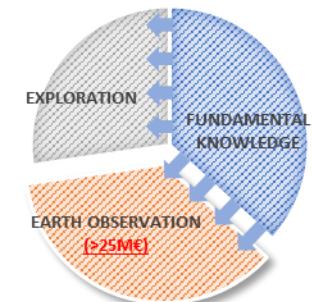
Nº OF PARTNERS



TOTAL BUDGET



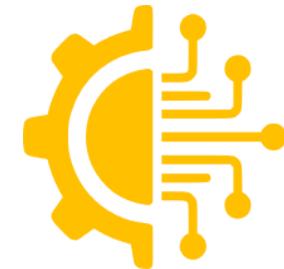
RESEARCH AREAS SHARE





OBJECTIVES

PROMOTE INNOVATION AND EXTEND FUNDAMENTAL KNOWLEDGE



FOSTERING A SUSTAINABLE FUTURE

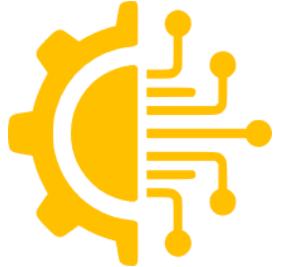


ENSURE LONG-TERM HUMAN PERMANENT IN EXTRATERRESTRIAL SPACE



STRENGTHENING THE 'ECOSYSTEM' SPACE IN ITALY

PROMOTE INNOVATION AND EXTEND FUNDAMENTAL KNOWLEDGE



Da sempre, **l'esplorazione e lo sfruttamento dello spazio** hanno richiesto di spingere la conoscenza oltre i propri limiti. SPACE IT UP svilupperà **tecnologie innovative** per sostenere e promuovere le future attività spaziali. I contributi di SPACE IT UP avranno un impatto su diverse aree fondamentali, come i modelli numerici, le architetture e le **costellazioni satellitari innovative**, i **nuovi profili di missione**, la strumentazione avanzata e le applicazioni basate sull'intelligenza artificiale.



FOSTERING A SUSTAINABLE FUTURE

L'umanità deve preservare il pianeta e lo spazio per le generazioni future. L'implementazione di tecnologie spaziali innovative consentirebbe di **osservare i cambiamenti climatici** e di prevedere gli eventi meteorologici estremi. Inoltre, SPACE IT UP proporrà soluzioni innovative per migliorare la **resilienza delle infrastrutture spaziali e terrestri ai gravi fenomeni meteorologici spaziali**.



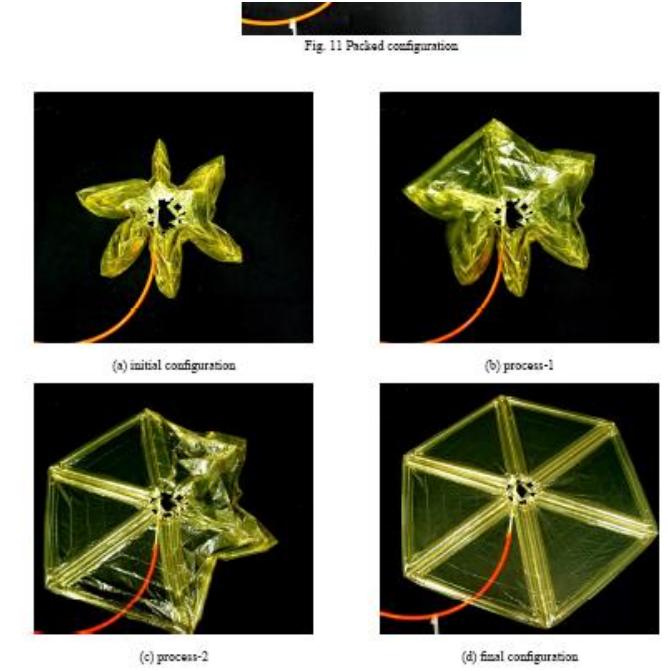
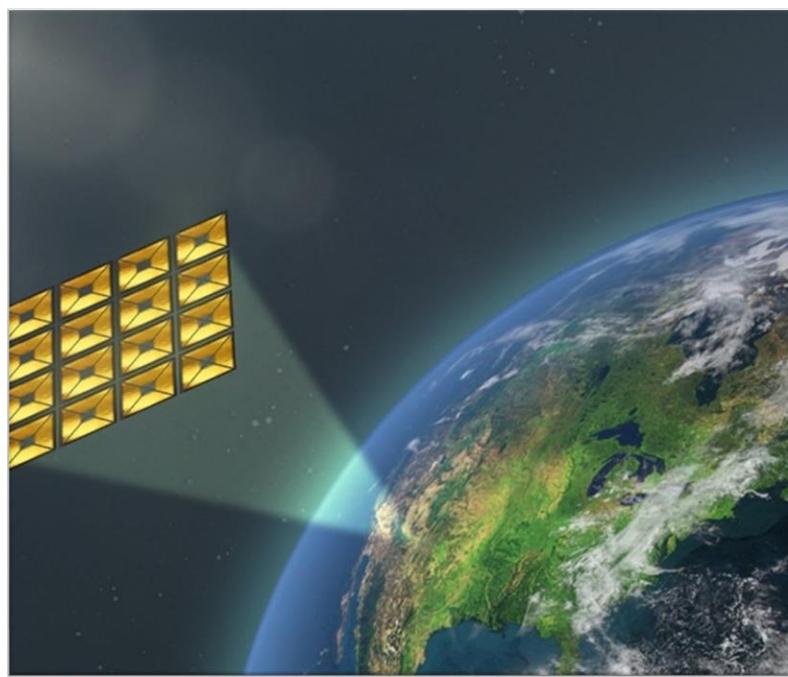
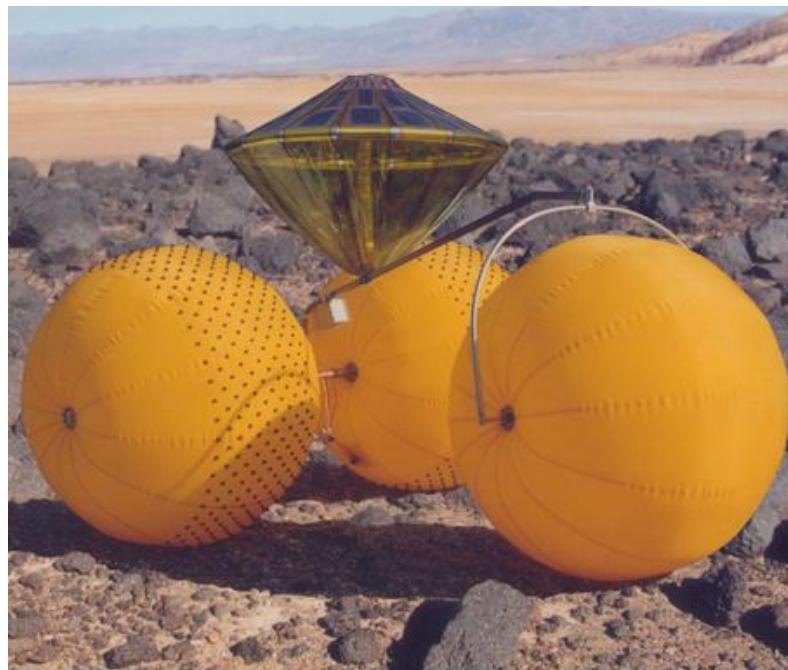
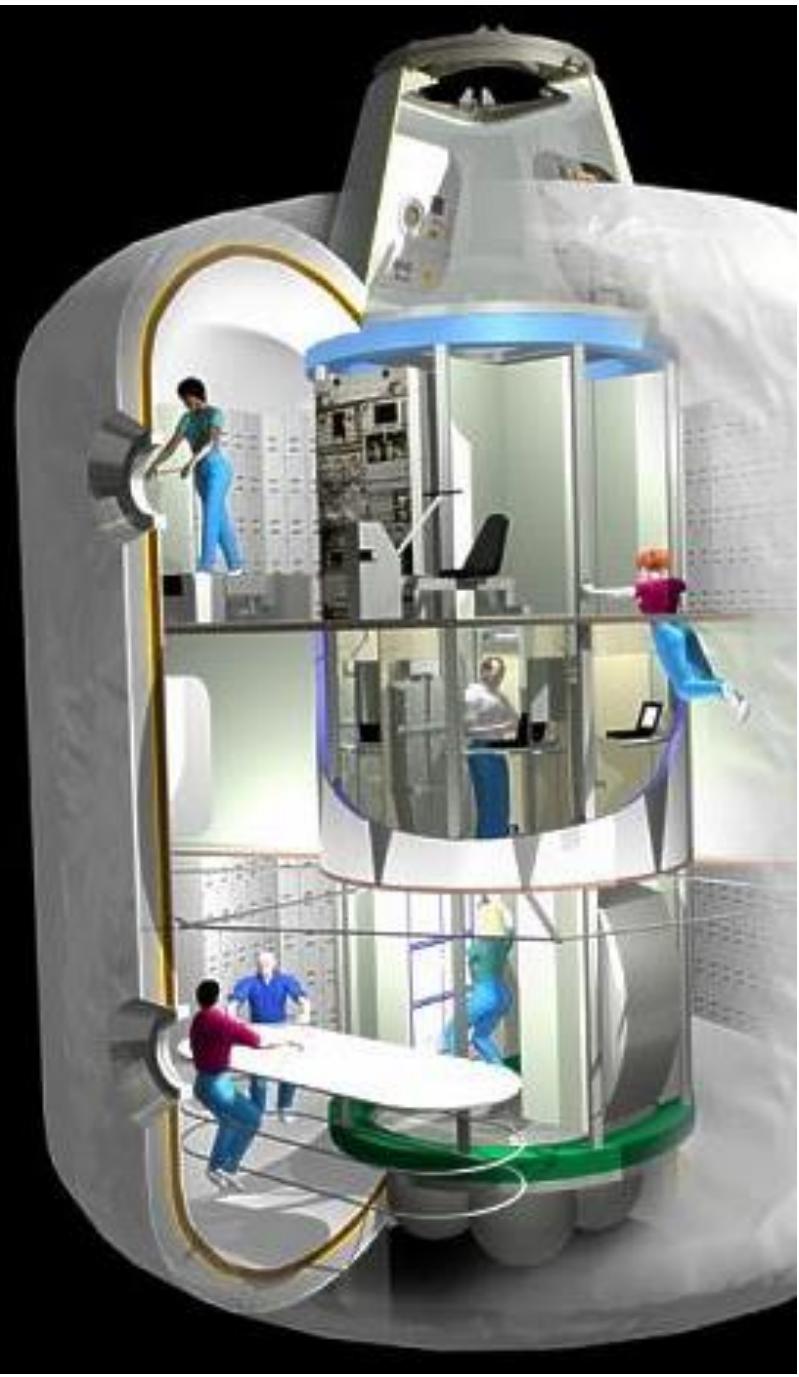
ENSURE LONG-TERM HUMAN PERMANENT IN EXTRATERRESTRIAL SPACE

La permanenza umana a lungo termine nello spazio pone numerose sfide tecnologiche che richiedono soluzioni innovative per essere superate. SPACE IT UP promuove lo sviluppo di **nuove idee e la definizione di tecnologie abilitanti** per rendere l'umanità una specie multiplanetaria. Il progetto affronterà non solo le questioni tecnologiche, ma anche quelle relative **allo sfruttamento delle risorse, alla produzione in situ, alle soluzioni circolari** per una permanenza sostenibile e agli aspetti neurofisiologici.



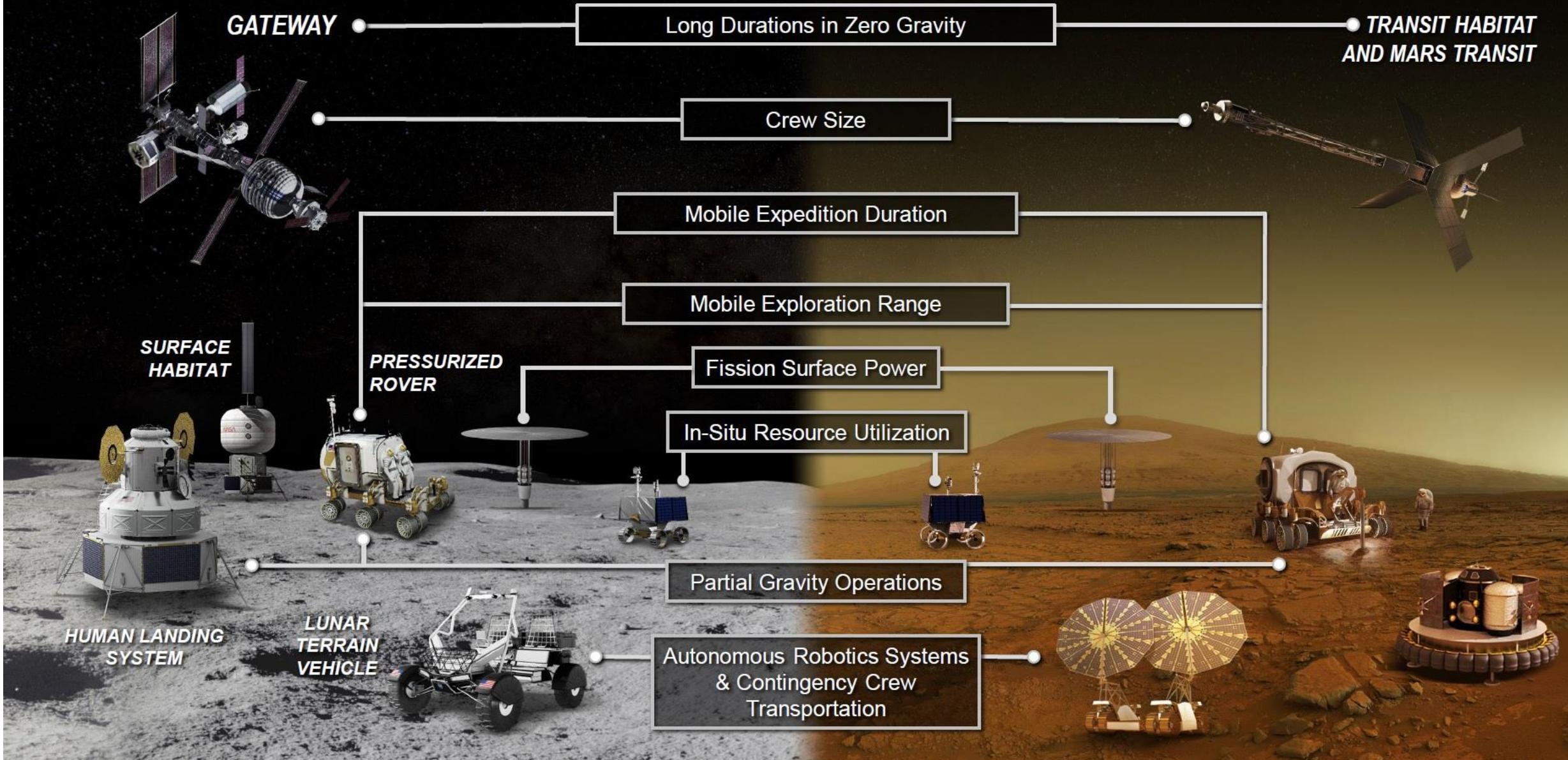
STRENGTHENING THE 'ECOSYSTEM' SPACE IN ITALY

L'Italia coprirà l'intera catena del valore della ricerca e dello sviluppo in campo spaziale, grazie a un efficace **coordinamento tra Università, Istituti di ricerca e un sistema di piccole, medie e grandi industrie**. Inoltre, SPACE IT UP promuoverà collaborazioni tra partner e nuove sinergie per proporre **soluzioni innovative e multidisciplinari**. Di conseguenza, SPACE IT UP **fornirà al Paese un ecosistema spaziale più solido e competitivo**.



Moon to Mars Exploration Strategy

Operations on and around the Moon will help prepare for the first human mission to Mars

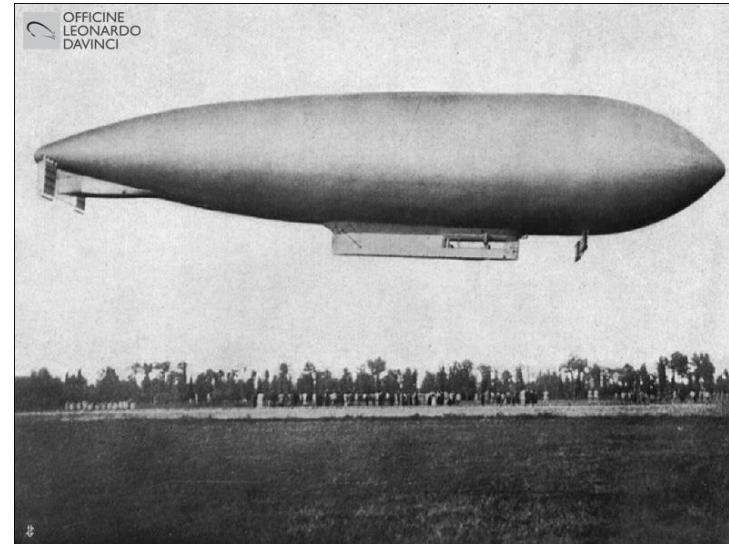




Contents

- **Gossamer and/or Inflatable**
- **Unmanned and Manned Application**
- **Complicating effects for Manned IS**
- **Example of Simulation of simple/complex (unmanned) elements**
- Conclusions**

INFLATABLE STRUCTURE VS AERONAUTICS:



[Condor_video](#)

Why Inflatable Space Structures?

- ❖ Strong Weight reduction is mandatory
- ❖ Less volume available

ARIANE 5 (Pay-load dimensions)	
Diameter of fairing	4,5 m
Height fairing	10,3 m
Max Payload	27 t

Possible solutions:

- Deployable structures DS
- Expandable structures ES
- Inflatable structures IS
- DS+ES+IS (Gossamer)

Complicating Effects!

- ❖ Use of IS as Manned Spacecraft





SIERRA
S P A C E

CONTENTS OF THIS TALK

- OVERVIEW OF INFLATABLE STRUCTURES FOR UNMANNED AND MANNED APPLICATIONS**

- EXAMPLE OF SIMULATION OF SIMPLE STRUCTURAL ELEMENTS AND COMPLETE SPACECRAFT (IRDT, IMOD, AIRLOCK ..)**

JUSTIFICATION OF THE INFLATABLE STRUCTURES

Many reasons are on the basis of the choice of inflatable structures for the construction of Space Structures

- 1. Mainly the mass of these structures is less of that of the traditional metallic ones**
- 2. The inflatable structures can be easily accommodated inside the launcher due to their possibility of folding**
- 3. Greater volumes can be exploited in comparison with the traditional metallic structures**
- 4. Different configurations can be easily obtained during their operative life (for example by the inflation of compartments in several periods)**

POSSIBLE UTILIZATION OF THE INFLATABLE STRUCTURES

IS can be classified as Unmanned and Manned Space Structures.

Example of Unmanned Space Structures are:

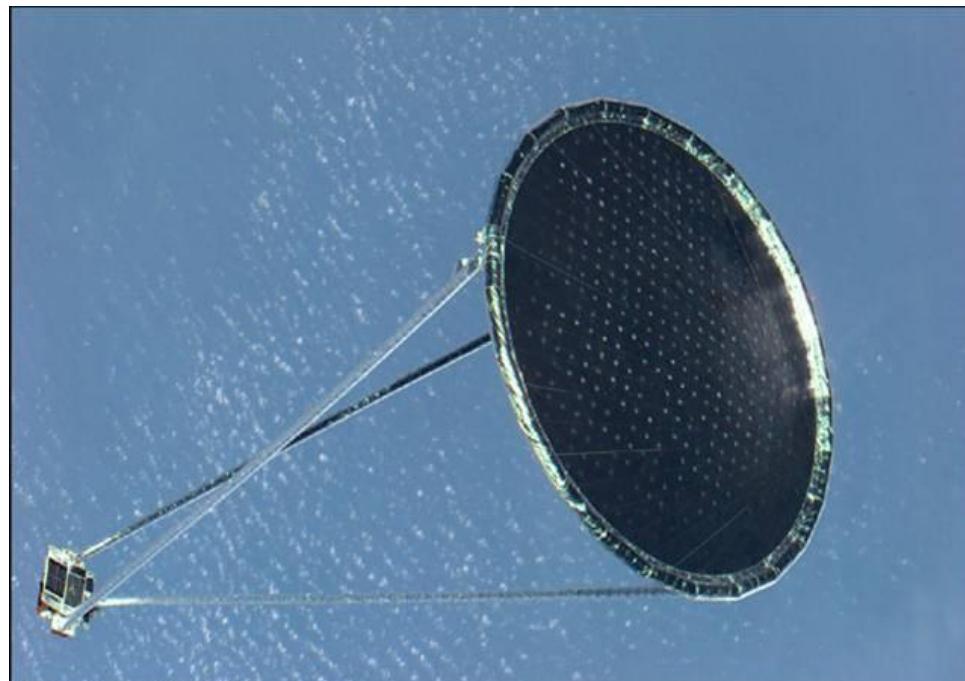
- **RE-ENTRY CAPSULES**
- **AIR BAGS FOR SHOCK ABSORBER AND DECELERATION SYSTEMS**
- **SOLAR ARRAYS**
- **ANTENNAS AND REFLECTORS**
- **HEAT SHIELDS**
- **SOLAR SAILS**
- **ROVER AND LANDER (R/L ELEMENTS)**

POSSIBLE UTILIZATION OF THE INFLATABLE STRUCTURES

Example of Manned Space Structures are:

- **SPACE STATION (!)**
- **PRESSURIZED MODULES PERMANENTLY ATTACHED
TO THE SPACE STATION**
- **PRESSURIZED MODULES FOR
INTER-PLANETARY TRANSFERRING**
- **PERMANENT BASES FOR PLANETARY EXPLORATION**

UNMANNED /INFLATABLE STRUCTURES



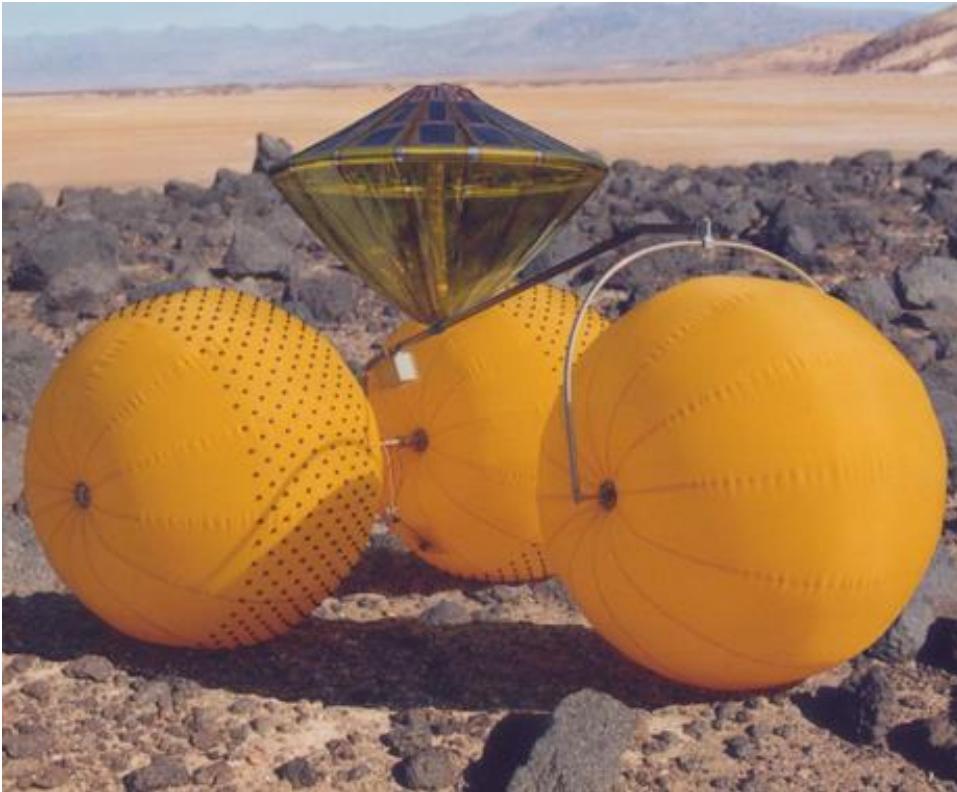
L'Garde Inflatable Antenna

**NORMALLY THESE ANTENNAS ARE LAUNCHED IN A ROLLED
CONFIGURATION AND THEN THEY ARE DEPLOYED IN ORBIT BY THE USE OF
NITROGEN**

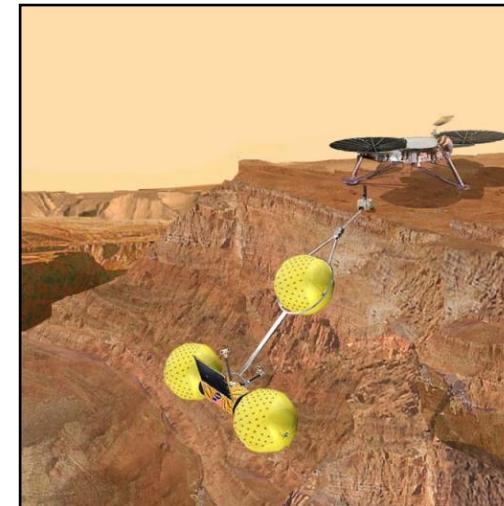
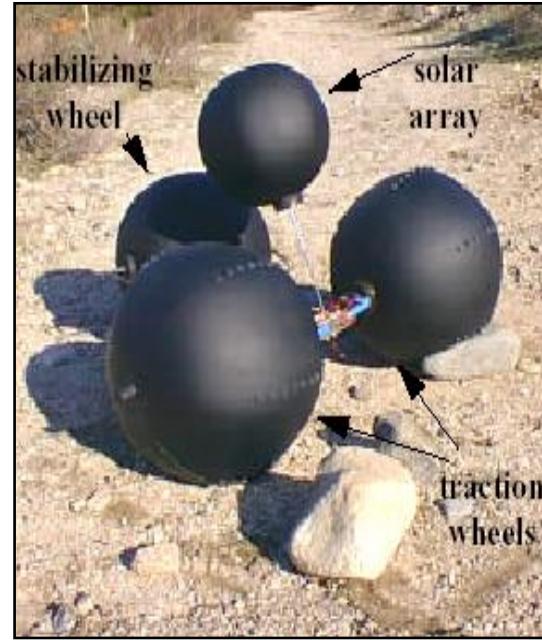
**THE REFLECTOR, WHICH IS MADE OF THIN LAYERS, IS TENSIONED BY THE
INFLATION OF THE TOROIDAL SUPPORT.**

EXAMPLE OF UNMANNED INFLATABLE STRUCTURES

Use of **Inflatable Capsule** for planetary exploration and for **Rover** construction



JPL Inflatable Rover per l'esplorazione della superficie di Marte alimentato da Inflatable Mars Rover Solar Array (ILC Dover, Inc.)



Constraints Introduced By Manned Configuration

1. The shell modules can become very thick due to the necessity of protection against the meteorites
2. The air containment become crucial for the crew survival
3. Then the connections between flexible and rigid parts become a critical topics, not only for the assurance of a suitable mechanical strength but also to avoid air leakage
4. The reaching of a precise operative shape can become sometimes mandatory, then deployable mechanisms can be used as guide to reach this final shape
5. The assurance of a suitable strength of this thick shell to the folding and unfolding stress, as well as to the internal pressure, become very important for the structure survival

Advantages Introduced By IS for Manned Configuration

Due to the fact that these structures must house the crew, the Manned Inflatable Space Modules can exhibit one of their main characteristic:

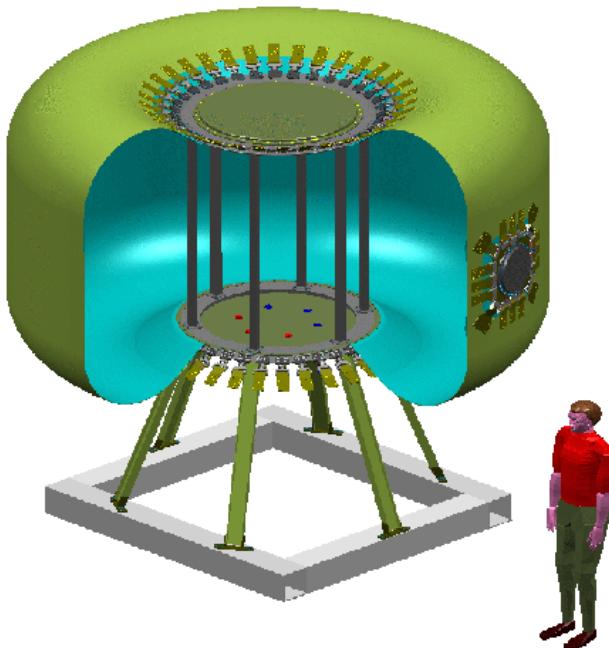
- 1. The exploitation of a large habitable volume**
- 2. The possibility of increasing the number of crew members and a more comfortable on board life**
- 3. A more comfortable internal environment, can give high benefit to the crew, especially for long duration missions**
- 1. The Manned Inflatable Modules can offer this large habitable, which can be 4 times greater then that offered by the traditional metallic modules.**

Manned Spacecraft



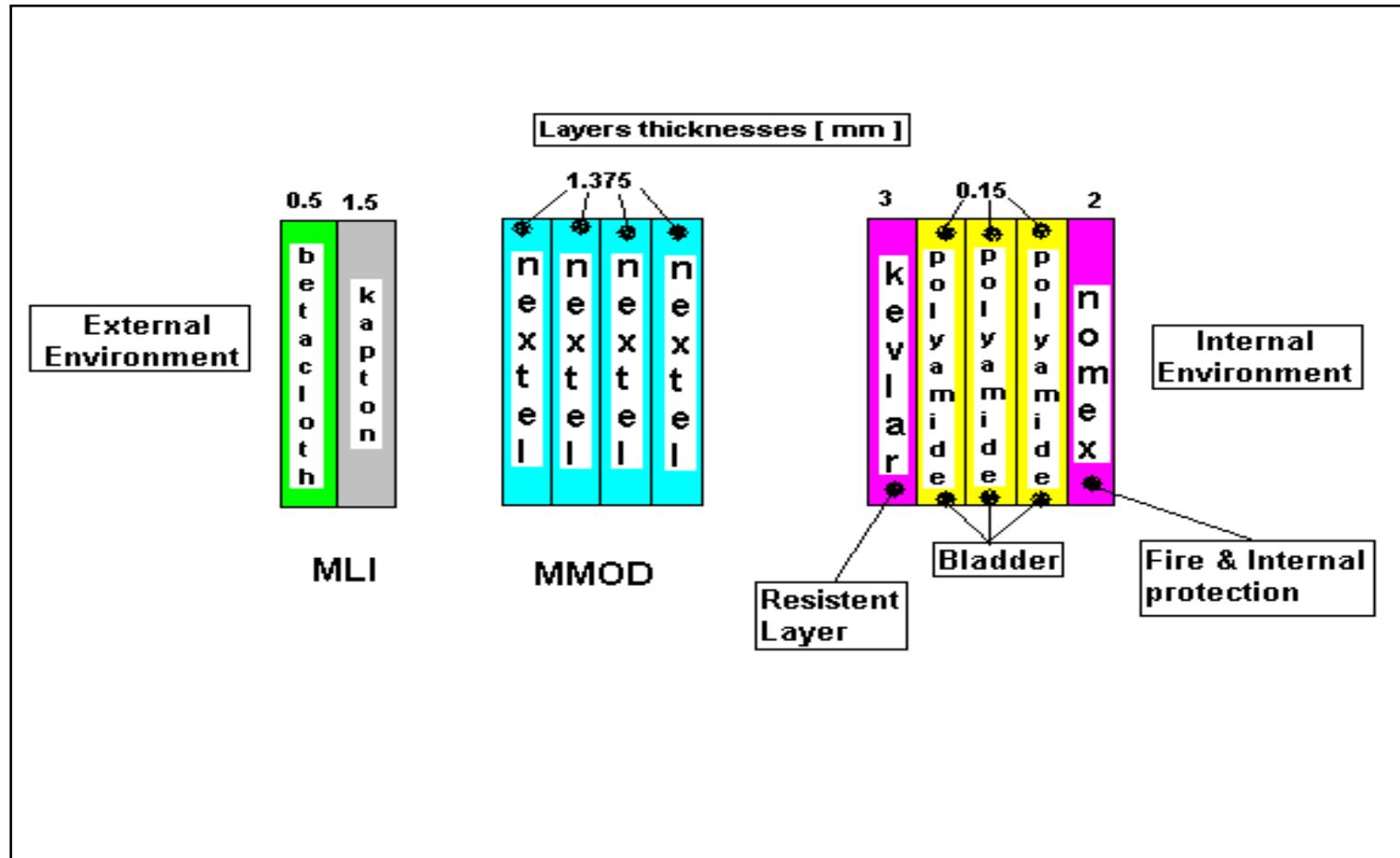
Tranhab –ISSA (NASA)

[Tranhab_video](#)



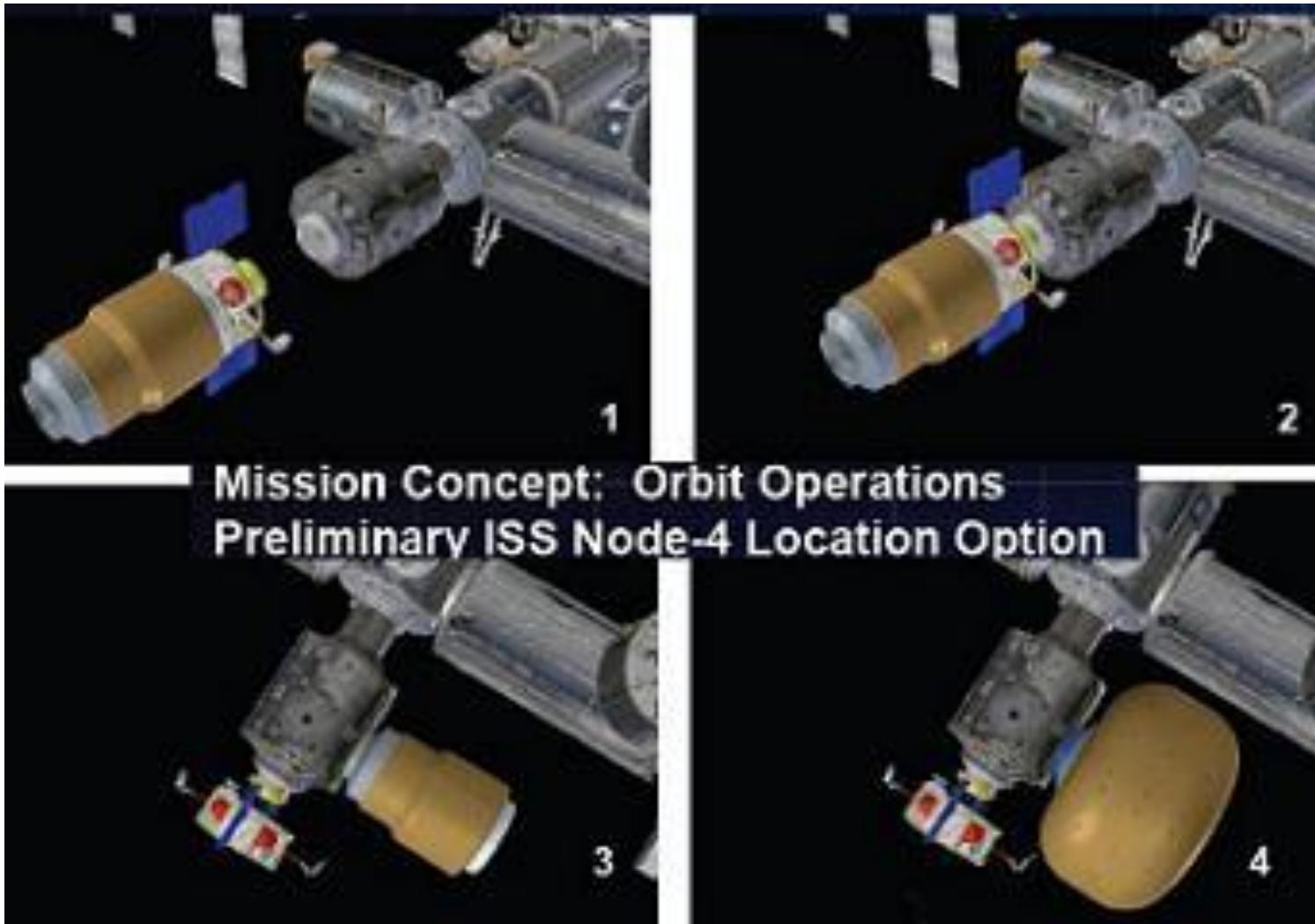
I-MOD (ESA, TASI)

Typical Multilayered Sequence



Inflatable Modules shell structure possible sequence

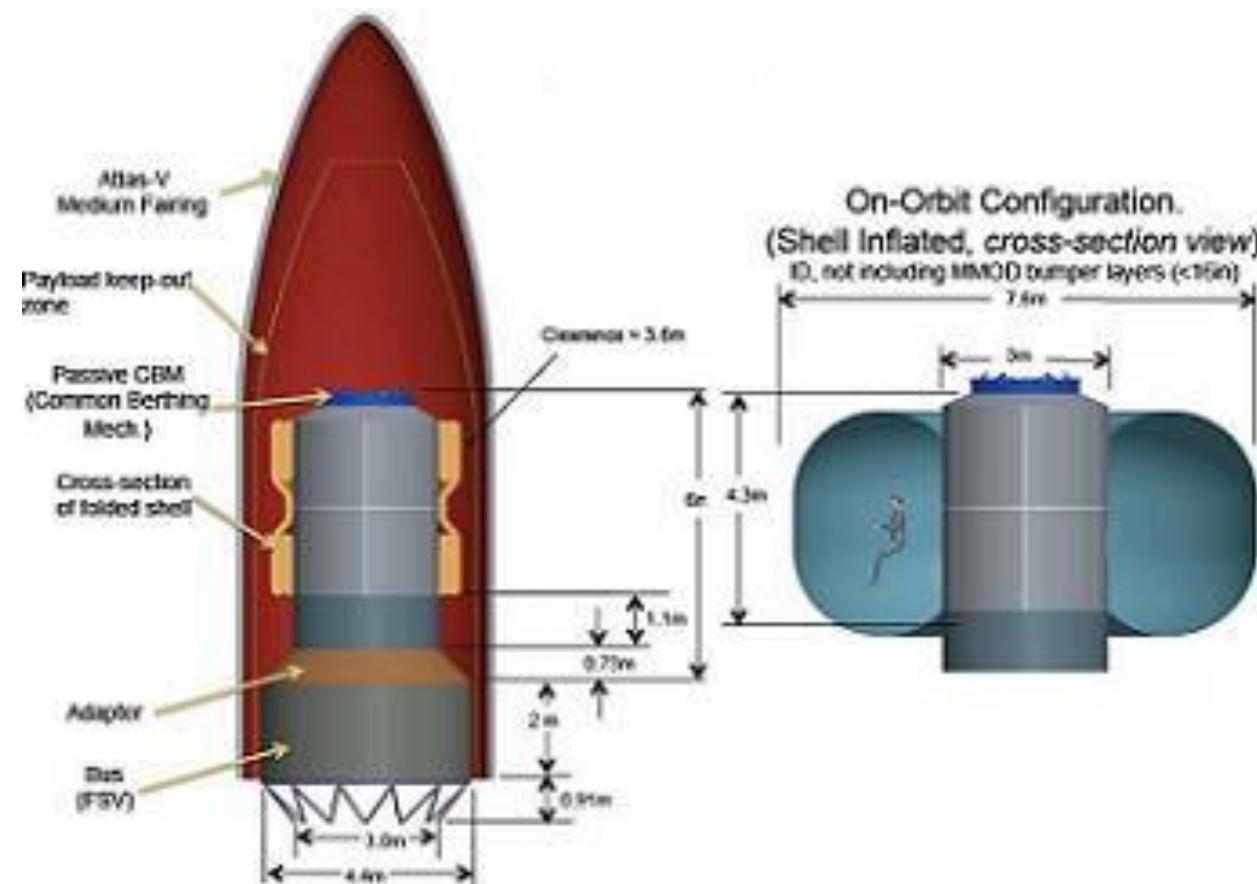
THE MOST RECENT GENESIS I & 2 FROM BIGELOW



Bigelow would provide the inflatable and inner core structure of the module, and perform all required flight analysis.

“An inflatable module has a rigid center core where the equipment is typically located and where the fabric is stowed for launch. After the module is berthed, it is inflated resulting in a pressurized fabric shell with a cylindrical core structure that houses equipment, etc.”

In-spite of their soft shell, Bigelow’s inflatable modules are more resistant to Micro Meteoroid Orbital Debris (MMOD) strikes than current metallic-shelled ISS modules



Beam @ ISS

- <https://www.designboom.com/technology/nasa-bigelow-beam-iss-space-habitat-module-12-06-2017/>

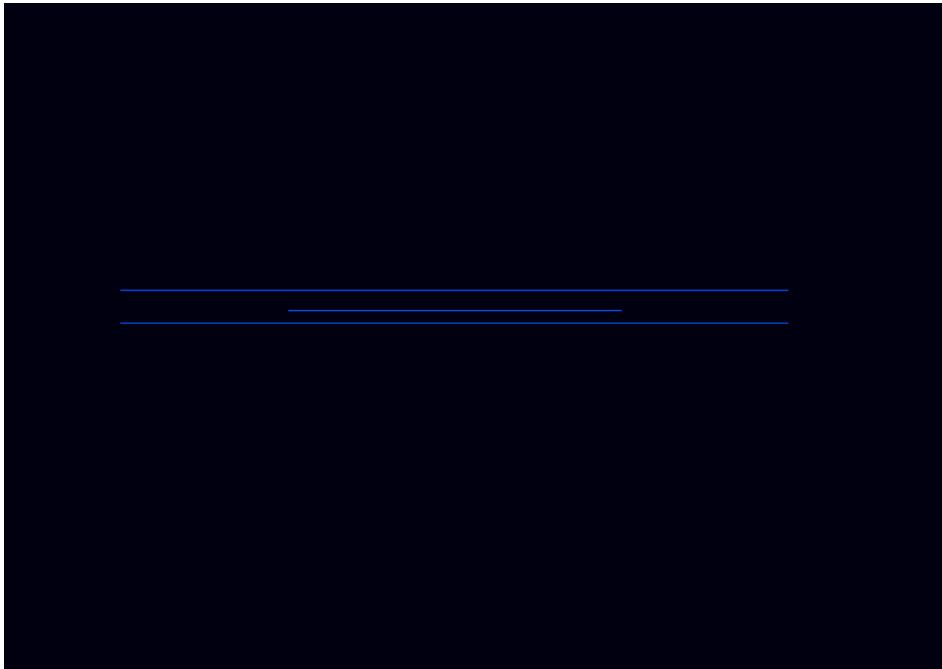
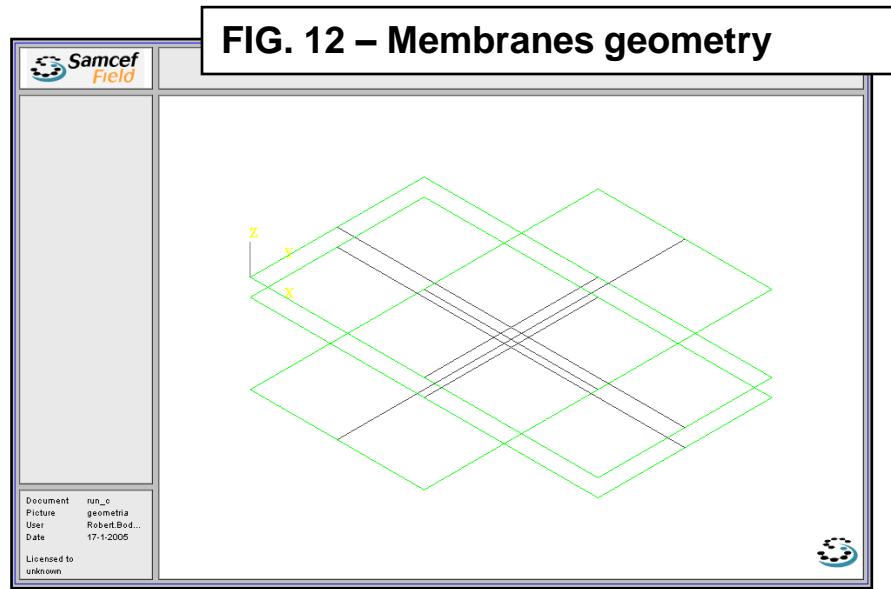
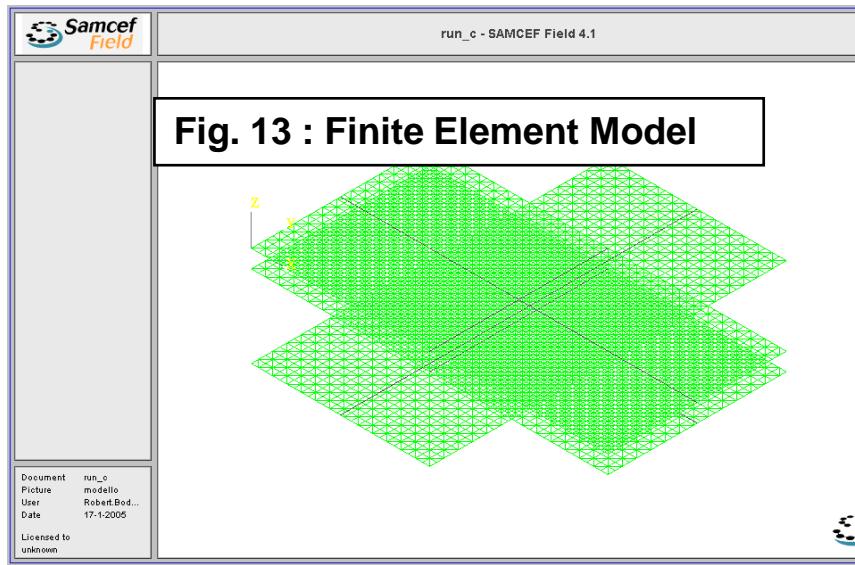
Simulations: Involved Disciplines

- Multibody Dynamics of rigid bodies**
- Multibody Dynamics of flexible bodies (FE)**
- Membrane flexibility and Membrane/bending Shell Theories**
- Layered Structures (manned!)**
- Inflation simulation (Large areas can be in contact)**
- Deployment simulation by inflation**
- Available codes (FE: Ls-dyna, Abaqus,)**
- Nonlinear phenomena are largely involved**
- Experiments are mandatory!**

THREE FLAT RECTANGULAR MEMBRANES

The following Figures 12 and 13 show respectively : the three membranes geometry and the

Finite Element Model. The applied uniform pressure load time history is a ramp from 0 to 1.



FLAT & TUBULAR MEMBRANE STRUCTURES

Next Figures 34 and 35 show the geometry of the tubes and of the flat shell respectively

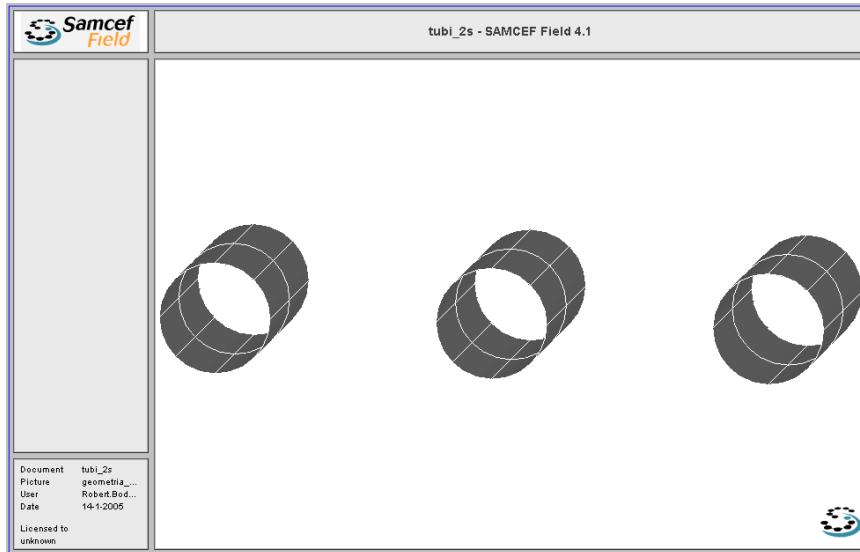


Fig. 34 :Geometry of the tubes

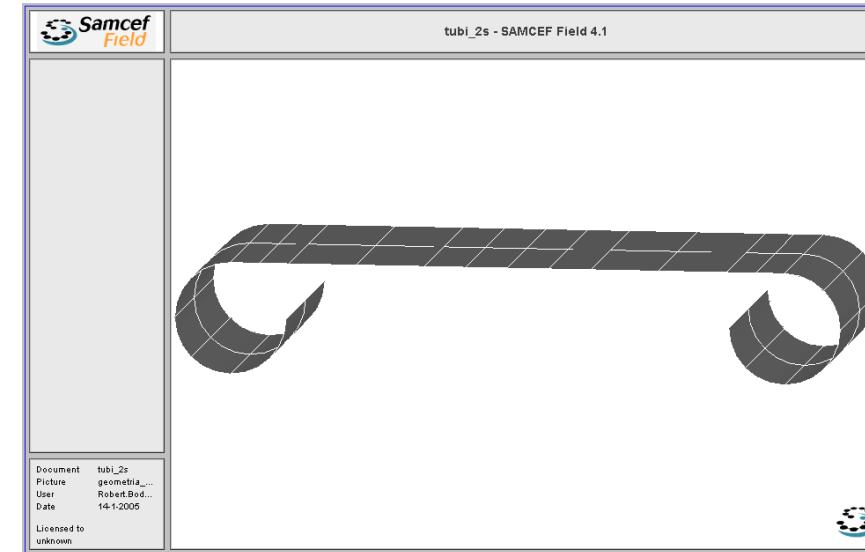


Fig. 35 :Geometry of the shell

FLAT & TUBULAR MEMBRANE STRUCTURES

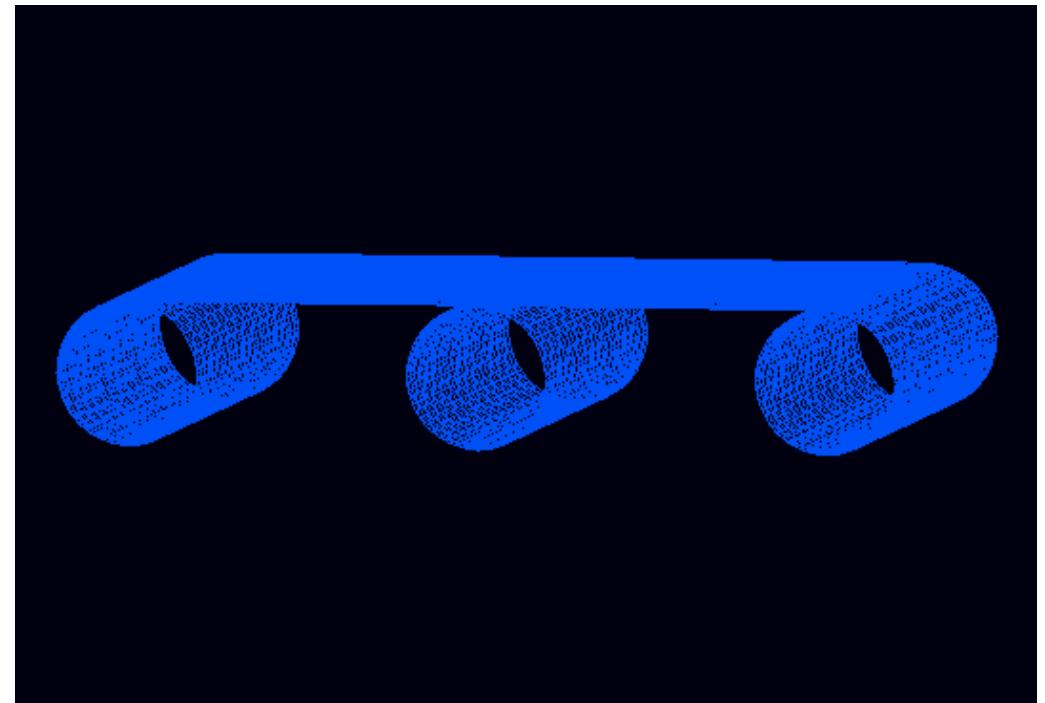
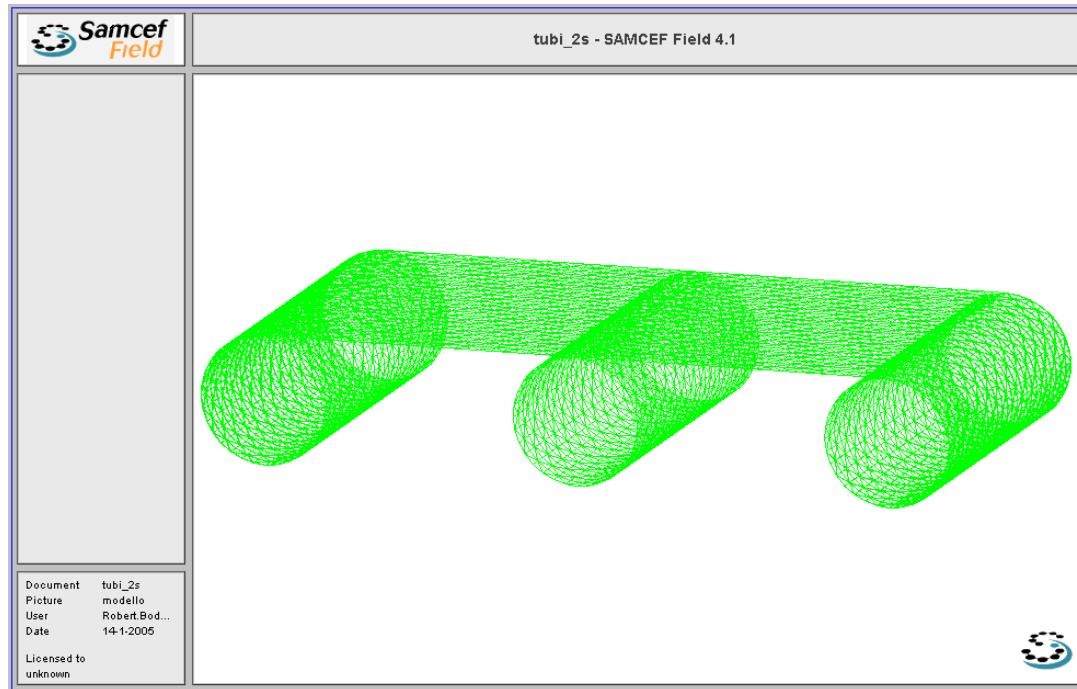


Fig. 36 :Finite Element Model

Example of Inflatable Re-entry capsule analysis

The CAD model of the capsule is shown in the next Figures 39, while the Figure 40 shows the 1/8 CAD model used to perform the analyses.

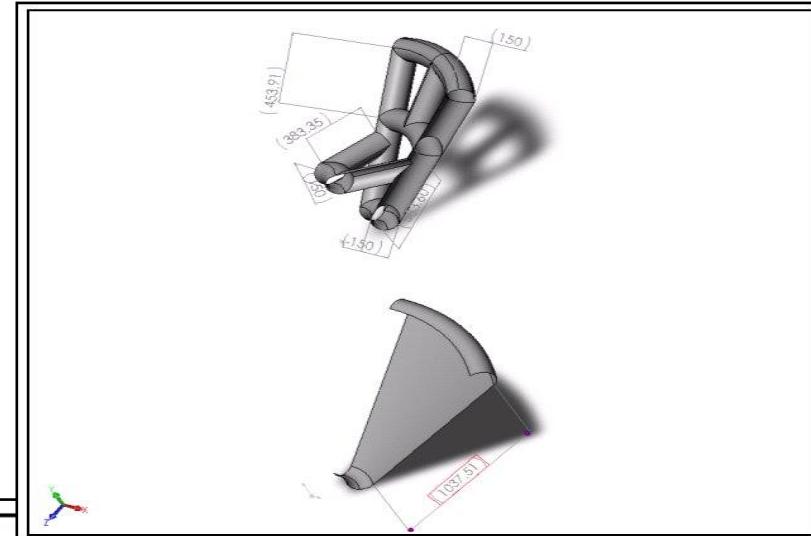
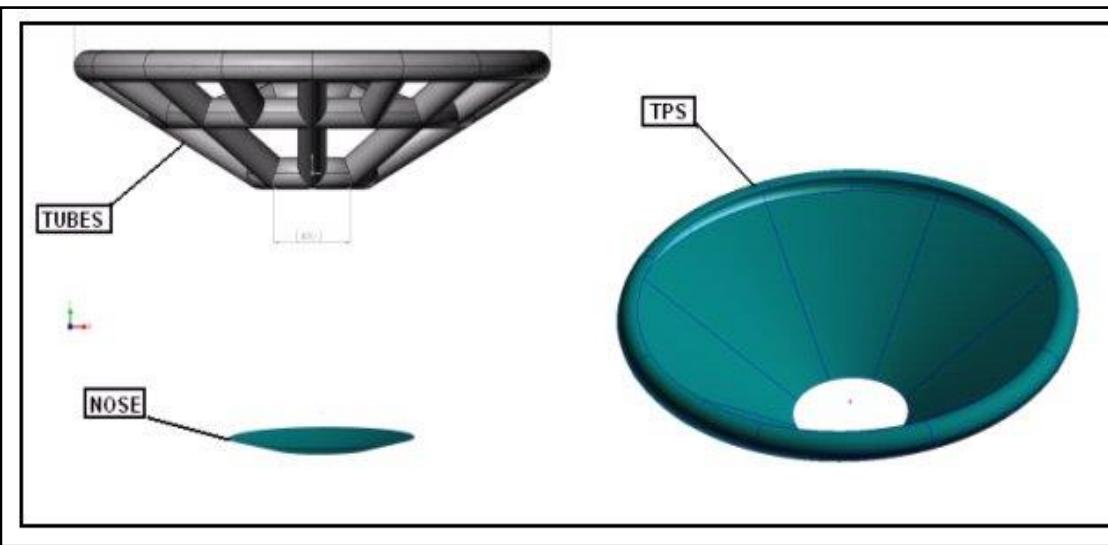
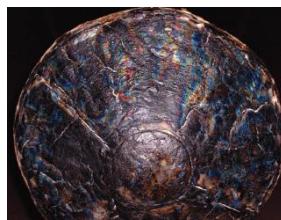


Fig. 40 :–1/8 CAD model – exploded view

Fig. 39 :– IRT CAD model – exploded view

EXAMPLE OF INFLATABLE RE-ENTRY CAPSULE ANALYSIS

- The results in term of displacements is shown in the next Figures 48

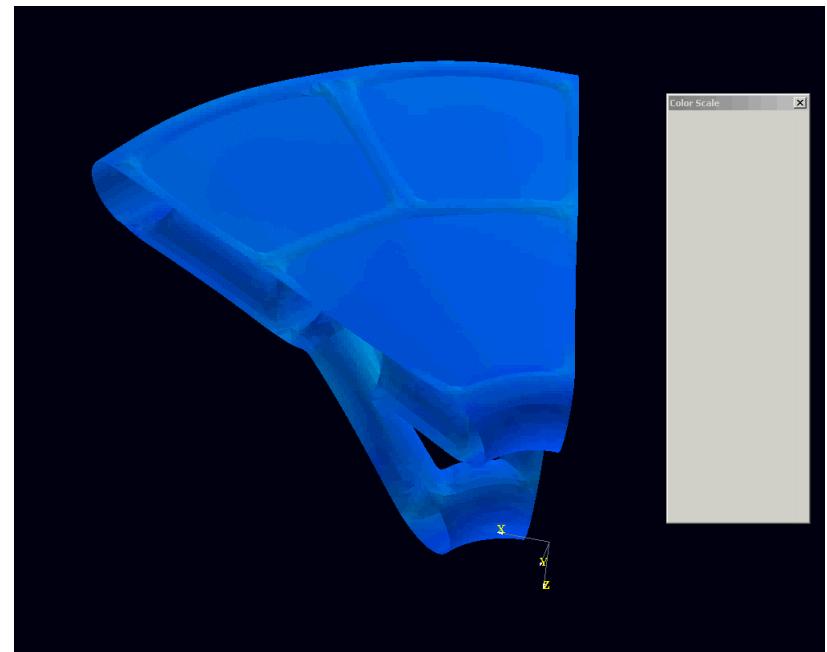
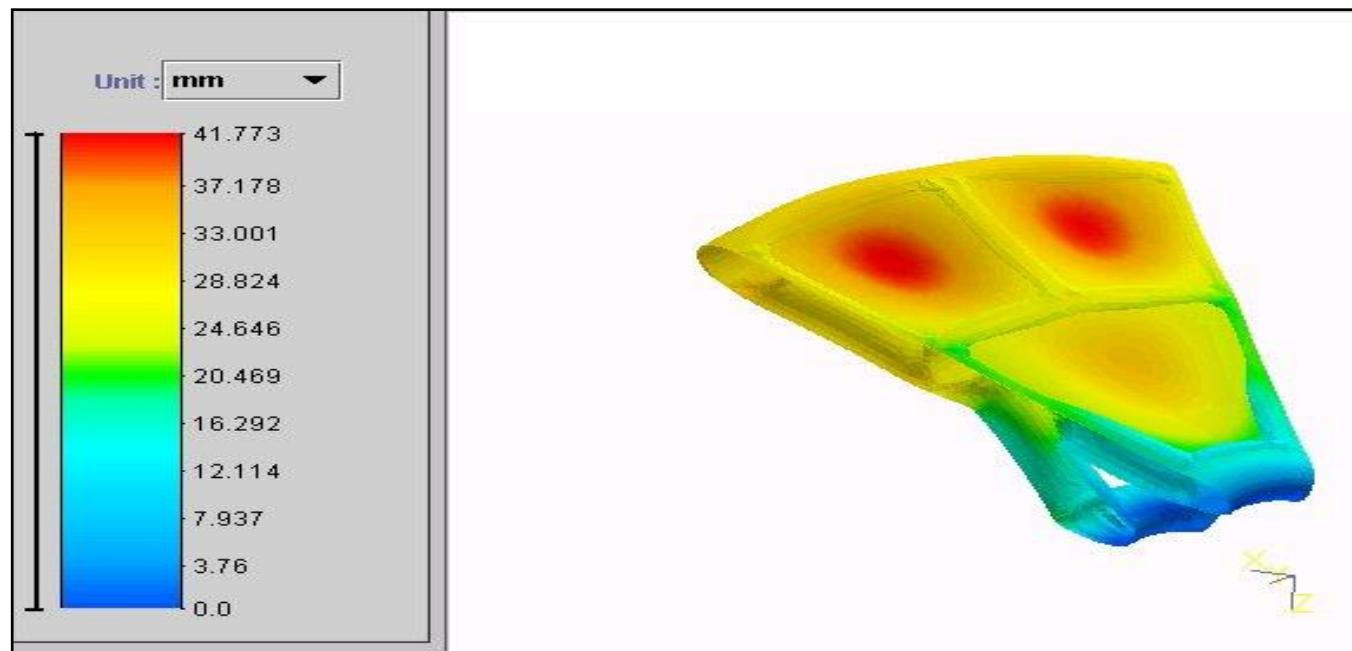
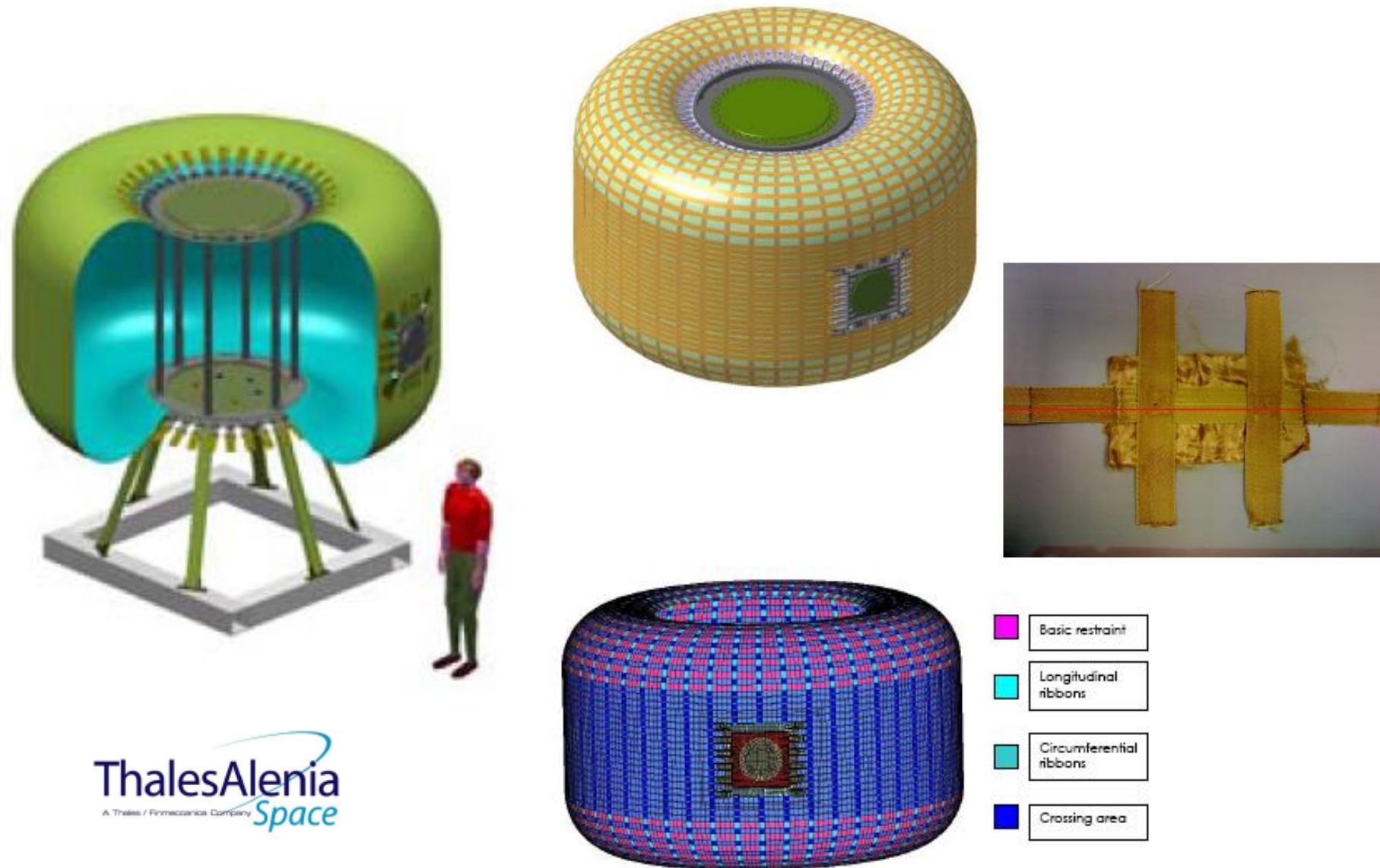


Fig. 48 – IRT displacements field

FE ANALYSY OF IMOD INFLATBALE MODULE

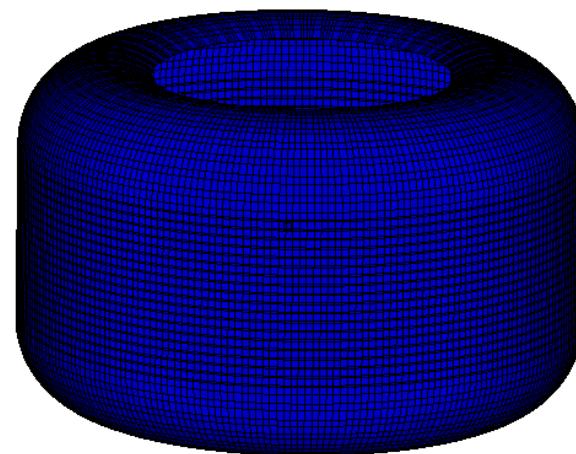
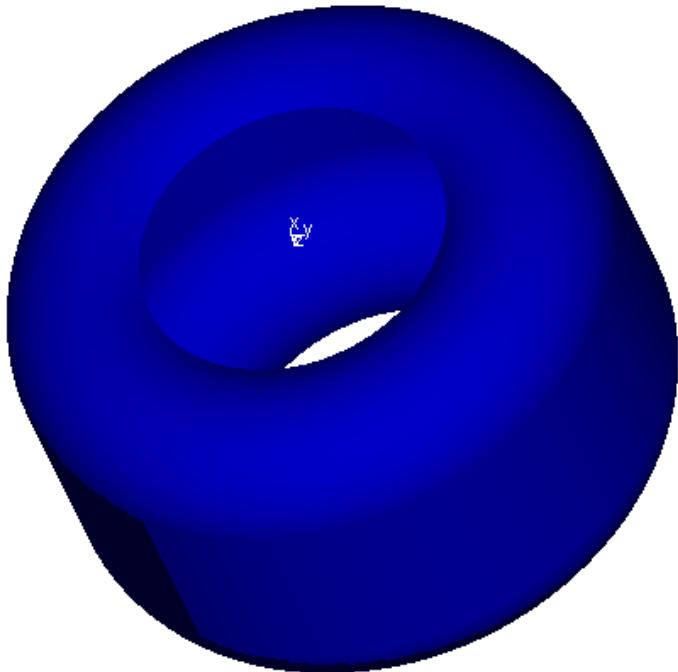
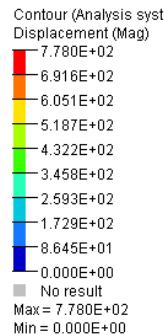


FE ANALYSY OF IMOD INFLATABLE MODULE

our Plot (Analysis system)
lacement(Mag)
1.915E+02
1.703E+02
1.490E+02
1.277E+02
1.064E+02
8.513E+01
6.384E+01
4.256E+01
2.128E+01
0.000E+00
No result
= 1.915E+02 (Node 74572)
= 0.000E+00 (Node 74499)



C:/Documents and Settings/marilisa.pischedda/Desktop/RISULTATI CORRETTI/PARTE I/ZYLON/zylon/zylon.k
Loadcase 1 : Time = 0.000000
Frame 1



Time = 0.000000

IMOD INFLATABLE MODULE

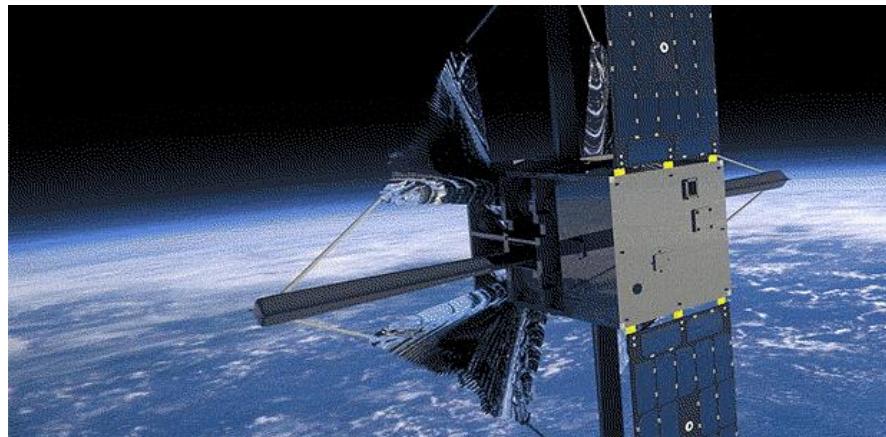


Deployable space booms – State-of-the-art (2) and current modelling issues



Two ways of deploying thin-shells structures:

Controlled



Advanced Composite Solar Sail System (ACS3) - Credits: NASA

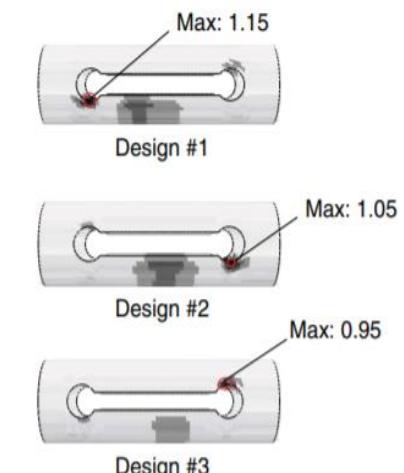
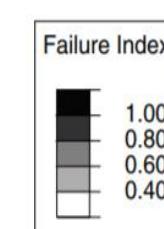
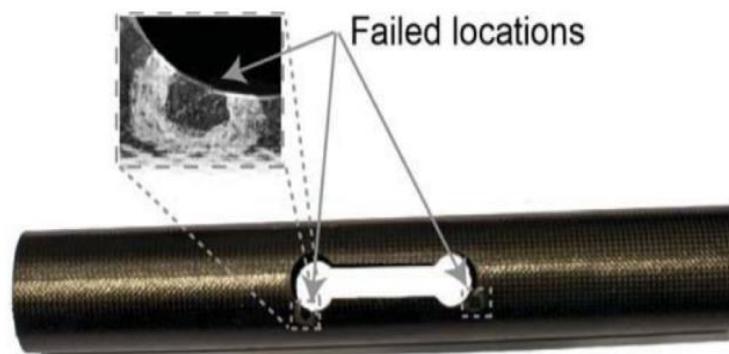
Unconstrained



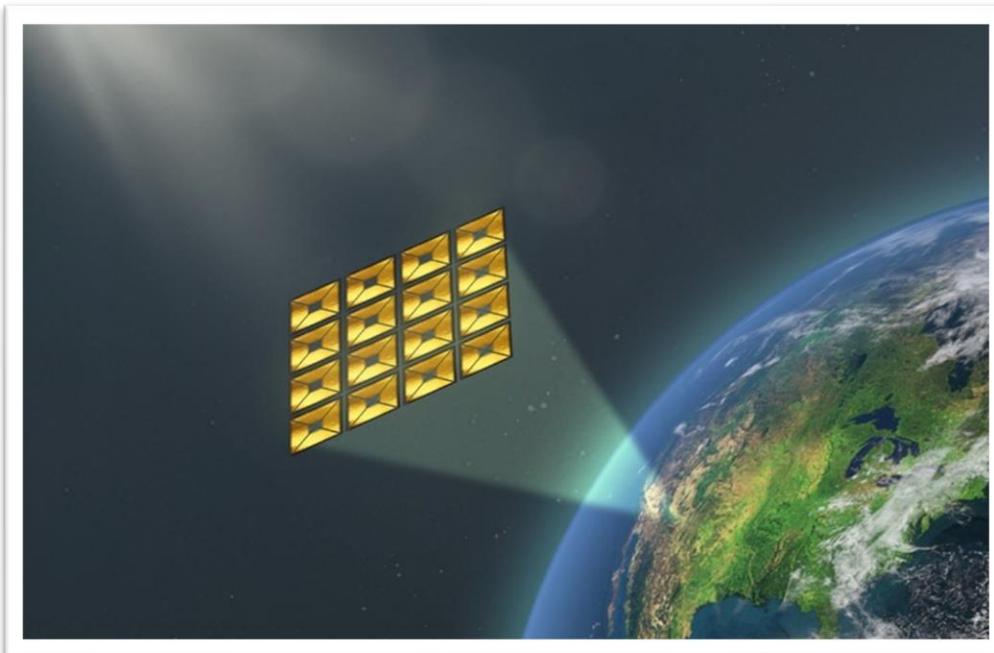
Wrapped rib reflector - credit: Oxford Space Systems

free-edge cracks
may occur

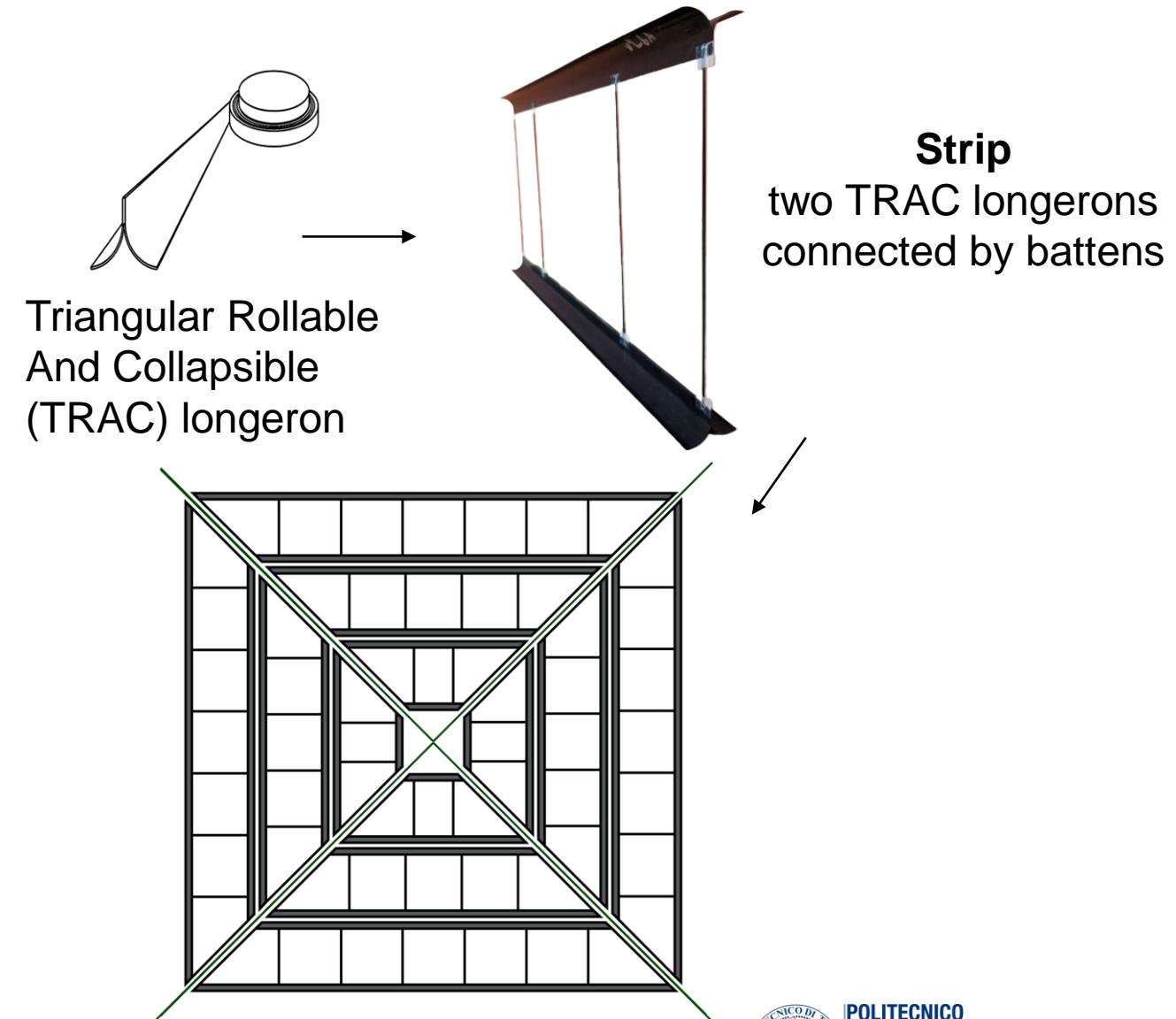
Mallikarachchi &
Pellegrino, *Journal of
Spacecraft and Rockets*,
2014.



Deployable space booms – Space Solar Power Project (1)



- Solution for harvesting solar energy in space
- Collects sunlight and wirelessly transmits power to Earth



CONCLUSIONS

- IS represent a valuable technique to be considered in next generation manned and un-manned space structures**
- IS are ‘probably’ the only solution to construct large spacecraft**
- Computational mechanics can be able to make simulation of Inflatable structures with some limitations (adequate capabilities are requested)**
- Available materials can be very much improved, dedicated one would be welcome (development on nanotech's could play a very significant role)**
- Experiments are mandatory**
- There are many ‘Ideas’ of FUTURE INFLATBALE spacecraft but difficult to implement**